

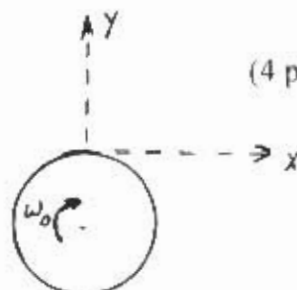
1998 US Physics Team, Exam 1
Solutions and Grading Key, Creative Response
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Concerning double jeopardy: Some problems require answers to be used in subsequent parts. Incorrect *answers* should be penalized only once. Correct *reasoning* should be acknowledged.

1. (25 points)

(a. 4) $\mathbf{v}_0 = R\omega_0 \mathbf{i}$ (in terms of the coordinate system shown)

(4 points)



(b. 7) This is a projectile problem: with $t = 0$ at the top of the disc, we have

$$x = v_0 t = R\omega_0 t \quad (2)$$

$$\text{and } y = -\frac{1}{2}gt^2. \quad (2)$$

$$\text{At point B, } y = -R, \text{ so that } t = \sqrt{2R/g} \quad (2)$$

$$\text{and therefore } x = R\omega_0 \sqrt{2R/g} \quad (1)$$

(c. 7) One may use either conservation of energy, or projectile kinematics.

Energy conservation approach:

$$E_A = E_B \quad (\text{recognizing the principle}) \quad (3)$$

$$\frac{1}{2}m(R\omega_0)^2 + mgR = \frac{1}{2}mv_B^2 + 0 \quad (\text{applying it}) \quad (2)$$

$$v_B = [(R\omega_0)^2 + 2gR]^{1/2} \quad (\text{solving}) \quad (2)$$

Projectile kinematics approach:

$$v_B^2 = v_x^2 + v_y^2 \quad (2)$$

$$\text{where } v_x = R\omega_0 \quad (2)$$

$$\text{and } v_y^2 = 2gR \quad [\text{from } v_y^2 = v_{0y}^2 + 2a_y(y - y_0)] \quad (2)$$

$$\text{and thus, } v_B = [(R\omega_0)^2 + 2gR]^{1/2} \quad (1)$$

(d. 7) By conservation of angular momentum,

$$L = L' \quad (\text{recognizing the principle}) \quad (2)$$

$$\text{which becomes } I_0\omega_0 = I'\omega' \quad (\text{applying it}) \quad (1)$$

Let's calculate I' : it has two contributions, one for the damaged disc missing the chip, and one for the chip: When the chip comes loose,

$$I' = [I_0 - mR^2] (\text{for the damaged disc}) + [mR^2] (\text{for the chip}) \quad (3)$$

$$= I_0.$$

$$\text{Hence, } \omega_0 = \omega'. \quad (1)$$

2. (25 points)

(a. 5) $\mathbf{F} = m\mathbf{a}$ becomes $kx_c - mg = 0$ (3)

so that $k = mg / x_c = (1 \text{ kg}) (9.8 \text{ N/kg}) / (0.2 \text{ m}) = 49 \text{ N/m}$. (2)

(b. 10) One may use the work-energy theorem,

$$W_{nc} = \Delta E \quad (3)$$

where W_{nc} is the work done by non-conservative forces and E is the total mechanical energy. With the zero of gravitational P.E. at the initial position of the hanging mass, $W_{nc} = \Delta E$ becomes

$$-\mu mgx = \frac{1}{2}kx^2 - mgx \quad (4)$$

The $x \neq 0$ solution gives $x = 2(mg/k)(1 - \mu) = 0.3 \text{ m}$. (3)

(c. 10) The system is equivalent to a particle of mass $2m$ being pulled in one direction by the force mg in the opposite direction by the force kx . Thus, Newton's second law gives

$$mg - kx = 2ma \quad (3)$$

which says $a + (k/2m)x = \frac{1}{2}g$, which is the equation of motion of a simple harmonic oscillator (in a uniform gravitational field), with angular frequency $\omega = \sqrt{k/2m}$. (4)

[Alternatively, if the student notes that the angular frequency squared equals k divided by the mass that is oscillating, and notes that $2m$ is the mass that oscillates, give full credit for part (c) up to here.]

Next, convert to period: $T = 2\pi/\omega$ gives $T = 2\pi\sqrt{2m/k} = 1.27 \text{ s}$. (3)

3. (25 points)

(a. 9) One may apply conservation of energy, $E_i = E_f$, (2)

and obtain (choosing P.F. = 0 at $\theta = 0$):

$$(2m)g(2L) = [\frac{1}{2}mv_1^2 + mg(2L - L\cos\theta)] + [\frac{1}{2}mv_2^2 + mg(2L)(1 - \cos\theta)]$$

which simplifies to

$$0 = \frac{1}{2}mv_1^2 + \frac{1}{2}mv_2^2 - 3mgL\cos\theta. \quad (3)$$

We note that for all θ , $v_1 = L\omega$ and $v_2 = (2L)\omega = 2v_1$. (2)

Hence, $v_1 = [(6/5)gL\cos\theta]^{1/2}$ and $v_2 = 2[(6/5)gL\cos\theta]^{1/2}$ (1)

so that, at $\theta = 0$, $v_1 = [(6/5)gL]^{1/2}$ and $v_2 = 2[(6/5)gL]^{1/2}$ (1)

(b. 16)

Consider the bottom mass (mass #2) [6 points to be earned here]:

The radial component of $\mathbf{F} = m\mathbf{a}$ becomes

$$T_2 - mg\cos\theta = m v_2^2 / 2L \quad (3)$$

Using the result $v_2^2 = (24/5)gL\cos\theta$ from part (a), one finds

$$T_2 = (17/5)mg\cos\theta \quad (2)$$

which is $< 4.2mg$ for all θ , so the lower rod does not break. (1)

[Note: if the student says "we need consider only the top rod because if either rod breaks it must be the top one," award the (1) point for this]



conclusion; however, the the student will still need to find T_2 as a function of θ in order to determine the angle where the upper rod breaks.]

Consider next the upper mass (mass #1) [10 points to be earned here]:

The radial component of $\mathbf{F} = m\mathbf{a}$ becomes

$$T_1 - T_2 - mg \cos \theta = mv_1^2/L \quad (5)$$

Using $v_1^2 = (6/5)gL \cos \theta$ and $T_2 = (17/5)mg \cos \theta$, one obtains

$$T_1 = (28/5)mg \cos \theta. \quad (3)$$

$$\text{Thus, } T_1 \text{ will equal } 4.2mg \text{ when } \theta = 41.4^\circ \quad (2)$$



4. (25 points)

(a. 7) $\mathbf{F} = m\mathbf{a}$ applied to the satellite gives

$$GMm/r^2 = mv^2/r \quad (3)$$

$$\text{where for the satellite, } v = 2\pi r/T. \quad (3)$$

Thus one obtains $T^2 = (4\pi^2/GM)r^3$,

$$\text{so that } C = 4\pi^2/GM. \quad (1)$$

(b. 6) For geosynchronous orbit, $T = 24 \text{ hrs.} = 8.64 \times 10^4 \text{ s.}$ (3)

$$\text{Thus Kepler's third law and the Earth data give } r = 4.22 \times 10^7 \text{ m.} \quad (3)$$

(c. 6) Let $T_0 = 8.64 \times 10^4 \text{ s.}$ and let $r_0 = 4.22 \times 10^7 \text{ m.}$ i.e., let the subscript denote the geosynchronous orbit data. With the change in orbit radius, we have

$r = r_0 + \Delta r$ and $T = T_0 + \Delta T$, so Kepler's third law becomes

$$T_0^2(1 + \Delta T/T_0)^2 = C r_0^3(1 + \Delta r/r_0)^3. \quad (1)$$

Use the binomial expansion to obtain

$$\Delta T = (3/2)(T_0/r_0)\Delta r. \quad (3)$$

$$\text{With } \Delta r = -2 \text{ km, it follows that } \Delta T = -6.1 \text{ s.} \quad (2)$$

[Of these last two points, one point is for the 6.1 s; one point is for either including the minus sign or stating that the satellite's period is *shorter* than before.]

(d. 6) Both the satellite and observer are revolving about the Earth's center of mass.

Let v_s = speed of the satellite relative to Earth's center of mass;

Let v_o = speed of the observer relative to Earth's center of mass.

The quantity we seek is $v_s - v_o$. Thus,

$$\begin{aligned} v_s - v_o &= R_e(\omega_s - \omega_o) \\ &= 2\pi R_e(1/T_s - 1/T_o) \end{aligned} \quad (2)$$

Writing $T_s = T_0(1 + \Delta T/T_0)$ and using the binomial expansion, one finds

$$v_s - v_o = -(2\pi R_e/T_0^2)\Delta T. \quad (2)$$

$$\begin{aligned} \text{Numerically, } v_s - v_o &= -2\pi(6.37 \times 10^6 \text{ m})(8.64 \times 10^4 \text{ s})^{-2}(-6.1 \text{ s}) \\ &= 3.3 \text{ cm/s.} \end{aligned} \quad (1)$$

Because the Earth rotates from west to east, and because the satellite completes one "lap" about the Earth's CM in 6.1 s less than the observer, the spot moves *east* of the observer at the speed of 3.3 cm/s. (1)